



U.S. Department
of Transportation
**Federal Aviation
Administration**

Rotorcraft Directorate
Fort Worth Manufacturing Inspection District Office
2601 Meacham Boulevard
Fort Worth, Texas 76137-4298

April 15, 2011

Superior Air Parts, Inc.
621 S. Royal Lane, Suite 100
Coppell, Texas 75019

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFACTURER APPROVAL

In accordance with the provisions of Title 14, Code of Federal Regulations (14 CFR), part 21, Certification Procedures for Products and Parts, subpart K, the Federal Aviation Administration (FAA) has found that the design data, based on Test and Computations submitted by Superior Air Parts, Inc., with your letters dated March 28, 2011 (FAAL-1445) and April 4, 2011 (FAAL-1447), meet the airworthiness requirements of the regulations applicable to the products on which the parts are to be installed. Additionally, the FAA has determined that Superior Air Parts, Inc., has established the fabrication inspection system required by § 21.303(h) at 621 S. Royal Lane, Suite 100, Coppell, Texas, 75019. Accordingly, Parts Manufacturer Approval (PMA) is hereby granted for production of the replacement parts listed in the enclosed Supplement No. 97, revision 2. Supplement No. 97, revision 2, dated April 15, 2011, updates Supplement No. 97 dated August 26, 1992. The purpose of the revision is to include over-sized and under-sized parts, not previously listed on Supplement No. 97. Supplement No. 97 remains in effect.

You are reminded that the provisions of 14 CFR, Parts 21 and 45, noted in our PMA letter of approval dated March 30, 2006, also apply to the enclosed PMA Listing-Supplement No. 97, revision 2. The enclosed supplement should be retained with the original PMA letter as evidence of approval to produce the parts concerned.

Sincerely,

A handwritten signature in blue ink that reads "Carlton N. Cochran". The signature is fluid and cursive, with a long horizontal stroke at the end.

Carlton N. Cochran
Manager, Fort Worth Manufacturing Inspection District Office
SW MIDO-42

Enclosure
PMA Supplement No. 97, revision 2



U.S. Department
of Transportation
**Federal Aviation
Administration**

FEDERAL AVIATION ADMINISTRATION – PARTS MANUFACTURER APPROVAL

SUPERIOR AIR PARTS, INC.
621 S. Royal Lane, Suite # 100
Coppell, TX 75019-3805

PMA NO. PQ0060SW
SUPPLEMENT NO. 97
REVISION 2
DATE April 15, 2011

Part Name	Part Number	Approved Replacement For Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Insert, Spark Plug Helical Coil	SA520112	520112	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Part Name	Part Number	Approved Replacement For Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Insert, Spark Plug Helical Coil	SA520112-P03	520112	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Insert, Spark Plug Helical Coil	SA520112-P05	520112	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Insert, Spark Plug Helical Coil	SA520112-P10	520112	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Part Name	Part Number	Approved Replacement For Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Insert, Spark Plug Helical Coil	SA520112-3	520112-3	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Insert, Spark Plug Helical Coil	SA520112-3-P03	520112-3	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Insert, Spark Plug Helical Coil	SA520112-3-P05	520112-3	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R

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Insert, Spark Plug Helical Coil	SA520112-3-P10	520112-3	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA520112, <u>REV:</u> A <u>DATE:</u> 09/30/1999 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F E165-2 E185-1, -3, -8, -9, -11 E225-4, -8 IO-346-A IO-360-A, -C, -D, -G, -H, -J, -K LTSIO-360-E, -KB TSIO-360-A, -B, -C, -D, -E, -F, -GB, -H, -JB, -KB, -LB, -MB IO-470-C O-470-A, -B, -E, -G, -J, -K, -L, -M, -P, -R, -S IO-470-G, -J, -K, -P, -R
Pushrod Assy.	SA537870	537870	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA537870, <u>REV:</u> F <u>DATE:</u> 04/11/2008 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F

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Part Name	Part Number	Approved Replacement For Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Pushrod Assy.	SA537870 P30	537870 P30	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA537870, <u>REV:</u> F <u>DATE:</u> 04/11/2008 or later FAA-approved revisions	Continental	A65-1, -3, -6, -6J, -7, -8, -8F, -8FJ, -8J, -9, -9F, -9FJ, -9J, -12, -12F, -12FJ, -12J, -14, -14F, -14FJ, -14J A75-3, -6, -6J, -8, -8F, -8FJ, -8J, -9, -9J C75-8, -12, -14, -16, -8F, -12F, -14F, -16F C85-8, -12, -14, -16, -8F, -12F, -14F, -16F C90-8, -12, -14, -16, -8F, -12F, -14F, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F
Camshaft	SA-643067	643067	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA643067, <u>REV:</u> G <u>DATE:</u> 12/05/2003 or later FAA-approved revisions	Continental	O-200-A, -B
Camshaft	SA-643067P20	643067 P20	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA643067, <u>REV:</u> G <u>DATE:</u> 12/05/2003 or later FAA-approved revisions	Continental	O-200-A, -B
Cylinder Stud Assy.	SA10200-A1	642917	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA10200-A1, <u>REV:</u> U <u>DATE:</u> 05/29/2007 or later FAA-approved revisions	Continental	C75-8, -8F, -8FH, -8FHJ, -8FJ, -12, -12F, -12FH, -12FHJ, -12FJ, -12J, -15, -15F C85-8, -8F, -8FH, -8FHJ, -8FJ, -12, -12F, -12FH, -12FHJ, -12FJ, -12J, -14F, -15, -15F C90-8F, -8FJ, -12F, -12FH, -12FJ, -12FP, -14F, -14FH, -14FJ, -16F O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D GO-300-A, -C, -D, -E, -F

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Cylinder Assy. With Valves	SA10200-A20	642917A12	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D
Cylinder Assy. With Valves	SA10200-A21	642917A2	Test and Computations per 14 CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	C75-8, -8F, -8FH, -8FHJ, -8FJ, -12, -12F, -12FH, -12FHJ, -12FJ, -12J, -15, -15F C85-8, -8F, -8FH, -8FHJ, -8FJ, -12, -12F, -12FH, -12FHJ, -12FJ, -12J, -14F, -15, -15F
Cylinder Assy. With Valves	SA10200-A22	642917A13	Test and Computation per 14CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	C90-8F, -8FJ, -12F, -12FH, -12FJ, -12FP, -14F, -14FH, -14FJ, -16F
Cylinder Assy. With Valves	SA10200-A23	642917A1	Test and Computation per 14CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	GO-300-A, -C, -D, -E, -F
Cylinder Assy. With Valves & Piston	SA10200-A20P	642917A12	Test and Computations per 14CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	O-200-A, -B C-125-2 C145-2, -2H O-300-A, -B, -C, -D
Cylinder Assy. With Valves & Piston	SA10200-A21P	642917A2	Test and Computations per 14CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	C75-8, -8F, -8FH, -8FHJ, -8FJ, -12, -12F, -12FH, -12FHJ, -12FJ, -12J, -15, -15F C85-8, -8F, -8FH, -8FHJ, -8FJ, -12, -12F, -12FH, -12FHJ, -12FJ, -12J, -14F, -15, -15F
Cylinder Assy. With Valves & Piston	SA10200-A22P	642917A13	Test and Computations per 14CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions	Continental	C90-8F, -8FJ, -12F, -12FH, -12FJ, -12FP, -14F, -14FH, -14FJ, -16F

FEDERAL AVIATION ADMINISTRATION – PARTS MANUFACTURER APPROVAL

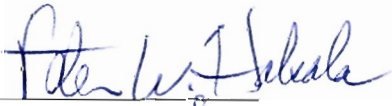
SUPERIOR AIR PARTS, INC.
 621 S. Royal Lane, Suite # 100
 Coppell, TX 75019-3805

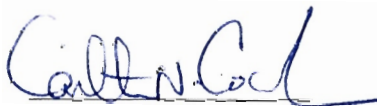
PMA NO. PQ0060SW
 SUPPLEMENT NO. 97
 REVISION 2
 DATE April 15, 2011

Part Name	Part Number	Approved Replacement For Part Number	Approval Basis and Approved Design Data	Make Eligibility	Model Eligibility
Cylinder Assy. With Valves & Piston	SA10200-A23P	642917A1	Test and Computations per 14CFR § 21.303, <u>DWG NO:</u> SA10200, <u>REV:</u> S <u>DATE:</u> 03/23/2011 or later FAA-approved revisions -----End of Listing-----	Continental	GO-300-A, -C, -D, -E, -F

NOTE:

- 1) Provide minor design changes in a manner as determined by the ACO. Handle major design changes to drawings and specifications in the same manner as that for an original FAA-PMA.
- 2) The FAA approved ICA for the above parts with their designs. These ICA may refer to those of the respective parts from the holders of type certificates. Otherwise, provide supplemental ICA for differences in the replacement parts. Make referral statements or supplemental ICA readily available per 14 CFR 21.50.
- 3) Revision number 2, dated 4/15/2011, updates supplement number 97 dated August 26, 1992. The purpose of the revision is to include over-sized and configuration part numbers not previously listed on supplement number 97. Supplement 97, dated August 26, 1992, remains in effect.


 S. Frances Cox
 Manager, Special Aircraft Certification Office
 ASW-190


 Carlton N. Cochran
 Manager, Fort Worth Manufacturing Inspection District Office
 SW MIDO-42